

## Introduction

Recently, much attention has focused on particular types of coating, aimed to reduce the skin friction drag in laminar or turbulent flows [1], with a variety of engineering applications. The working mechanism is based on their superhydrophobic behavior, which is well known in Nature, since the description of the *lotus effect* by Barthlott and Neinhuis [2].



Figure 1: A lotus leaf showing a super-hydrophobic behavior, together with details of wall texture at microscope.

The key feature is the presence of a micro- or nano-structured surface that traps the air into small pockets over which the water can flow with low friction. The persistence of this condition, called Cassie-Baxter state, is as crucial as difficult to maintain in time since the gas layer can easily be depleted. The aim of this research is to develop a computation model, based on a multiscale approach, in order to characterize the drag reduction induced by superhydrophobic surfaces.

## The multiscale approach

The study is carried out by solving:

- One microscopic problem, which describes the flow in the proximity of the wall protrusions. The goal is to compute the **protrusion heights**  $\lambda_{||}$  and  $\lambda_{\perp}$ , which quantify the effectiveness of the wall structures in terms of drag reduction.
- One macroscopic problem, which evaluates the effect of superhydrophobic surfaces by imposing homogenized boundary conditions at the walls of a test channel. The boundary conditions require the protrusion heights values to be derived from the microscopic problem.
- In order to avoid mathematical and technical difficulties, the wall texture is assumed periodic along the spanwise direction and the interface is assumed to evolve slowly in  $x$ .

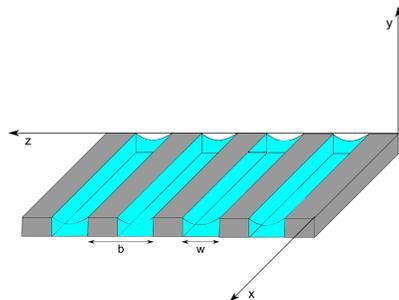


Figure 2: Typical surface corrugation studied.  $b$  is the spanwise periodicity of the wall pattern and  $w$  is the thickness of a micro-channel.

## The microscopic problem

The problem is governed by the Stokes equation, which can be decoupled into two parts, called **transverse** and **longitudinal** problem

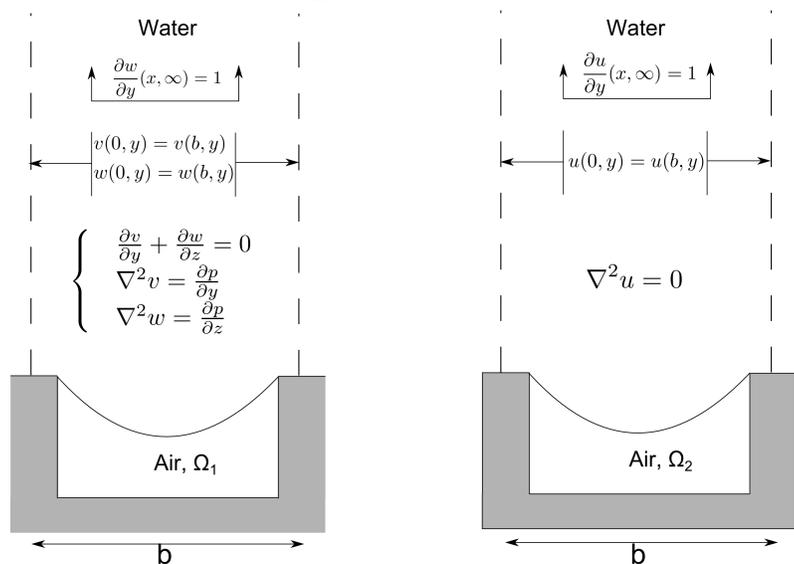


Figure 3: Geometry, governing equations and boundary conditions for the transverse (center) and the longitudinal (right) problems.

## References

- J.P. Rothstein and R.J. Daniello, Superhydrophobic Surfaces for Drag Reduction, provisional U.S. patent application 61177453, filed May 12, 2009.
- W. Barthlott, C. Neinhuis, Purity of the sacred lotus, or escape from contamination in biological surfaces. *Planta*, vol 202, pp 1-8, 1997.
- C. Pozrikidis, Boundary integral and singularity methods for linearized viscous flow. *Cambridge University Press*, 1992

## Boundary integral formulation

We use the boundary integral method to solve for the transverse and the longitudinal problems. The velocity field in a generic domain may be reconstructed using only the values of the velocity,  $\mathbf{u}$ , and stress fields,  $\mathbf{f}$ , on the closed boundary of the domain. This can be done by introducing two integral operators, called single-layer,  $\mathcal{F}^{SLP}$ , and double-layer,  $\mathcal{F}^{DLP}$ , potentials [3]. After mathematical manipulation, the governing equations can be recast in integral form as

$$\begin{cases} \alpha u_j(x_0) = -\mathcal{F}_j^{SLP}(x_0, \mathbf{f}; W) - \mathcal{F}_j^{SLP}(x_0, \mathbf{f}; T) + \hat{\mathcal{F}}_j^{DLP}(x_0, \mathbf{u}; T) \\ \quad - \mathcal{F}_j^{SLP}(x_0, \Delta \mathbf{f}; l) + (\lambda - 1) \hat{\mathcal{F}}_j^{DLP}(x_0, \mathbf{u}; l) \\ \alpha u(x_0) = -\lambda \mathcal{F}^{SLP}(x_0, \nabla \mathbf{u} \cdot \mathbf{n}; W_3) + \hat{\mathcal{F}}^{DLP}(x_0, \mathbf{u}; T) \\ \quad - \mathcal{F}^{SLP}(x_0, \nabla \mathbf{u} \cdot \mathbf{n}; W_1 + W_2 + T) + (\lambda - 1) \hat{\mathcal{F}}^{DLP}(x_0, \mathbf{u}; l) \end{cases}$$

with:

$$\begin{cases} \alpha = \frac{1+\lambda}{2}, & \text{if } x_0 \in l, \\ \alpha = \frac{1}{2}, & \text{if } x_0 \in T \cup W_1 \cup W_2, \\ \alpha = \frac{\lambda}{2}, & \text{if } x_0 \in W_3, \\ \alpha = \lambda, & \text{if } x_0 \in \Omega_2 / \{T + l + L + R + W_1 + W_2\}, \\ \alpha = 1, & \text{if } x_0 \in \Omega_1 / \{W_3 + l\}. \end{cases}$$

## Solution in a microscopic unit cell

We calculate the value of protrusion heights for different rigidity of the fluid interface, expressed in terms of the capillary number  $Ca = \frac{\mu u_\tau}{\sigma}$  and different air fractions trapped into the cavity.

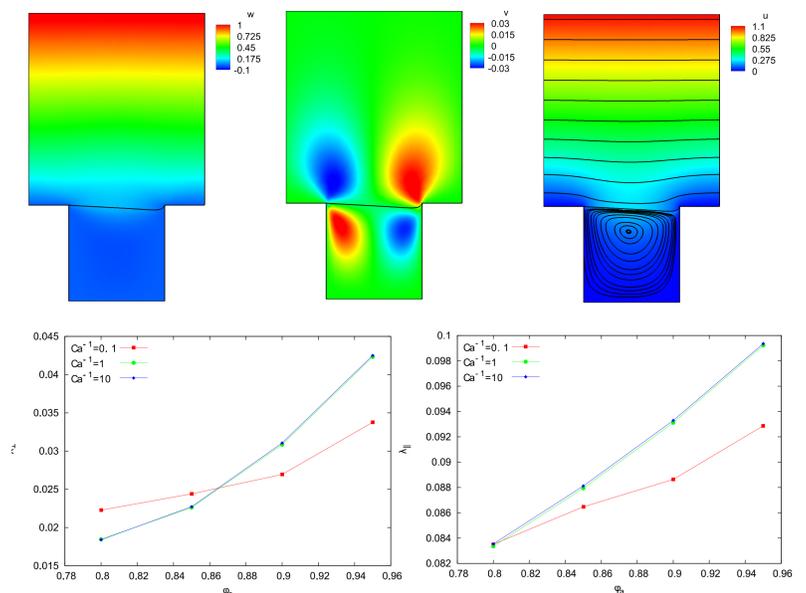


Figure 4: Computed velocity field for a flow over super-hydrophobic surfaces and associated protrusion heights for different air fractions and capillary numbers.

## The macroscopic problem

Direct numerical simulations in a standard channel of dimension  $6H \times 2H \times 3H$  are employed to measure the drag reduction induced by the superhydrophobic coating. The boundary conditions at the channel lower wall reads

$$\begin{cases} u = \tilde{\lambda}_{||} \frac{\partial u}{\partial y}, \quad \tilde{\lambda}_{||} = \frac{b}{H} \lambda_{||} \\ v = 0, \\ w = \tilde{\lambda}_{\perp} \frac{\partial w}{\partial y}, \quad \tilde{\lambda}_{\perp} = \frac{b}{H} \lambda_{\perp}. \end{cases}$$

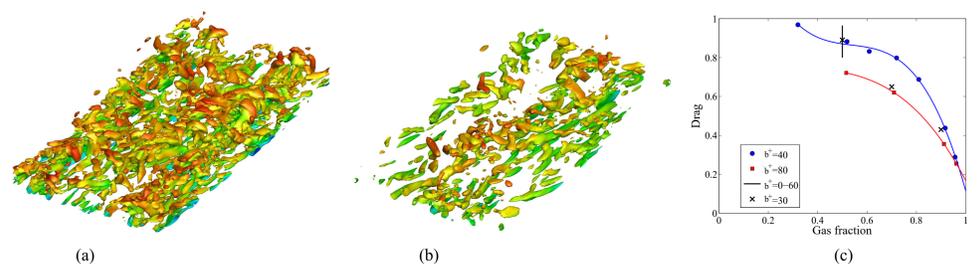


Figure 5: (a-b) Iso-surfaces of the Q-criterion colored by the streamwise velocity component; (c) Comparison between DNS and experiments by Park et al. [4]. Blue and red lines represent the least square fit of the experimental data; black solid line is the range of values achieved at  $GF = \frac{w}{b} = 0.5$  for different values of the spanwise texture periodicity in wall unit  $b^+$ . Crosses are additional simulations at different values of  $GF$ .

## Acknowledgment

The activities on superhydrophobic coatings have started thanks to a gratefully acknowledged Fincantieri Innovation Challenge grant, monitored by Cetena S.p.A.